WP9 - Numerical capability development for liquid icing conditions

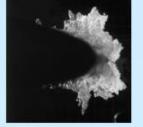


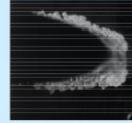
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AIRBUS SAS BOMB **CIRA** CU DASSAV ONERA – O. Rouzaud POLIMI POLYMO SAFRAN TUBS TUDA TUS

Context

- Objectives: improve and validate current 2D/3D numerical tools with respect to Appendix C and **Appendix O** conditions, so that they can be used for both design and certification of aircraft, rotorcraft and engines
- Why is it so important to work on SLDs?
 - Because large droplets do not behave as smaller droplets: more inertial, more energetic, ...
 - Need for adapted or specific physical models







DVM 100 µm



- Why is it so important to work on the numerical tools?
 - To improve the overall performances of the industrial solvers in 3D
 - To improve the solution by itself
- * Appendix C is associated to clouds / "small" cloud droplets
- * Appendix Dris associated to Freezing Drizzle and Freezing Rain conditions / Supercooled Large Droplets (SLD) ICE GENESIS Public Workshop - 3rd November 2022

Work plan

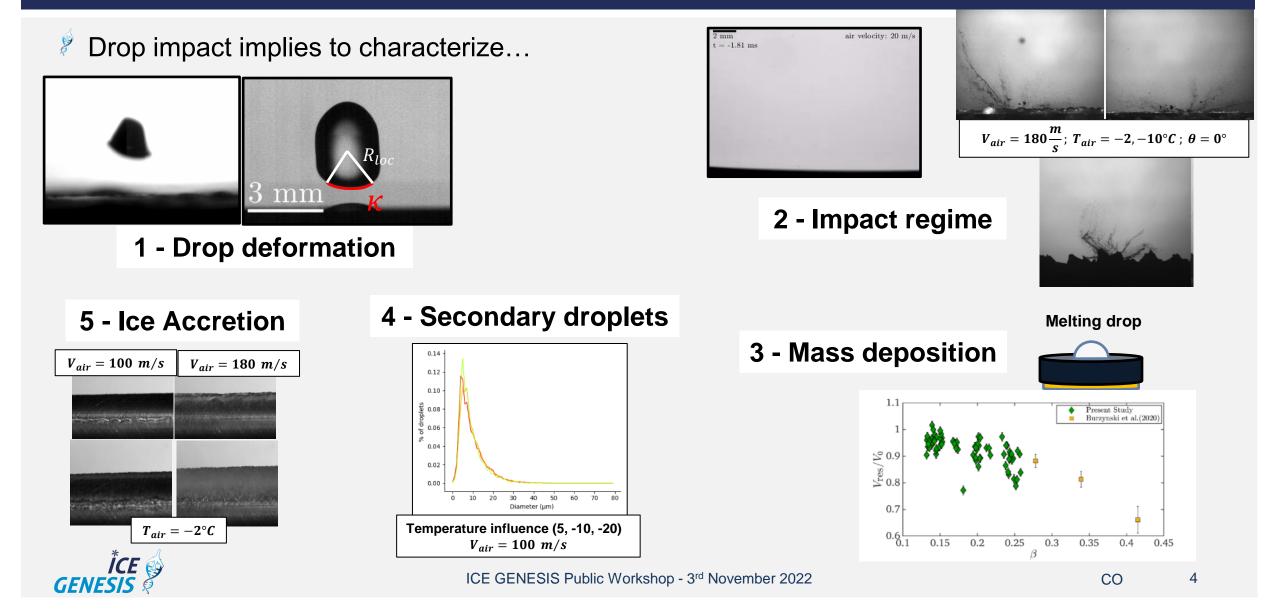
- Decomposition into 4 Tasks
 - Task 9.1: Basic experiments to provide missing data for model development
 - Task 9.2: Model improvements and implementation in 2D tools for calibration and preliminary validation
 - Phenomena under consideration: drop impact, ice roughness, liquid film runback
 - More or less academic experiments performed in different labs (CU, ONERA, TUDA, TUBS)
 - o Improvement or development of new physical models (CIRA, ONERA, POLIMI, TUBS, TUDA, TUS)

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- Task 9.4: Improvement of 3D ice accretion numerical methodologies (CIRA, ONERA, POLIMI, POLYMO)
 - Working on numerical models for meshing (automatic meshing, remeshing) for 3D test cases
- Task 9.3: Model integration in 3D numerical tools and preliminary validation
 - Combining physical models and numerical methods to answer Ice Genesis WP9 objectives
 - Towards the industrial configurations of WP11

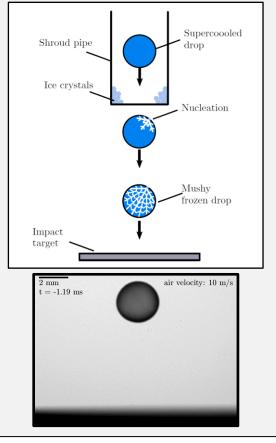


Drop Impact – Experimental activities

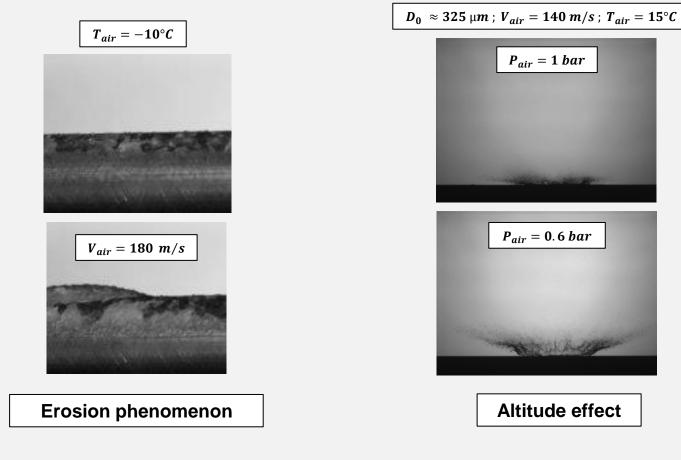


Drop Impact – Experimental activities

And also new or unexpected phenomena



Dendritically frozen drop impact



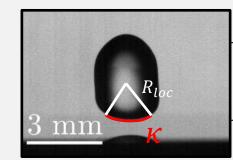


Drop Impact – Physical modelling

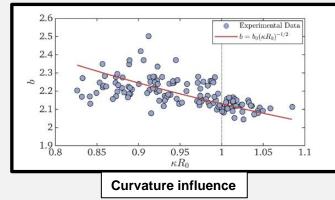
- From very detailed models...
 - Droplet deformation prior to the impact

• Spherical droplet:
$$r_{R_o} = b \sqrt{t U_o}_{R_o} \Rightarrow r^+ = b \sqrt{t^+}$$

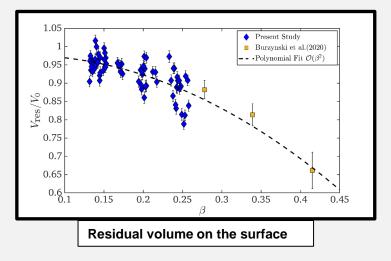
• Analysis of the IG data provides: $b = b_o / \sqrt{\kappa R_o}$, $b_o = 2.12$



r radius of the wetted area R_o radius of the impinging drop U_o drop velocity



 \Rightarrow influence on the splashing parameter β (Riboux-Gordillo model)

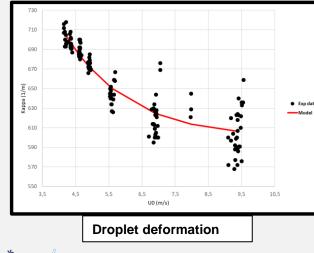


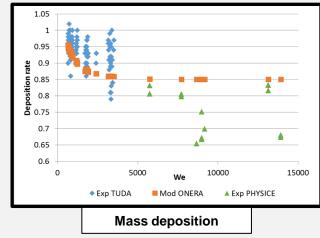
- Mass deposition
 - $\circ\,$ Depending on the splashing parameter $\beta\,$
 - Flight conditions may exceed $\beta > 0.4$ (what happens above 0.45?)
- Liquid film runback (not presented here)



Drop Impact – Physical modelling

- To applied ones used in the industrial solvers
 - Droplet deformation => how to define the radius of curvature κ ?
 - Mass deposition
 - o Adaptation of the Trontin-Villedieu and the Wright models on the impact function
 - o Application to an accretion experimental test case (Ice Genesis database)
 - Secondary droplets
 - Approach based on Riboux-Gordillo and Burzinsky-Bansmer-Roisman models
 - Description of the spray by a log-normal law defined by parameters estimated from RG



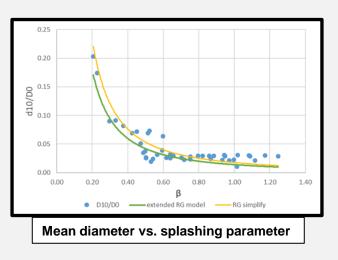


 $\kappa(U_0) = C_0 * [C_1 + C_2 \tanh(C_3 U_0)]$ $\kappa = \min(\kappa(U_0), 715)$

ONERA model:
$$f(\widetilde{K}_n) = 1 - C_1 \frac{\widetilde{K}_n^2}{C_2 + \widetilde{K}_n^2}$$

Wright model:

$$f_m = \frac{m_S}{m_0} = C_3 [1 - \sin(\theta_0)] [1 - e^{-0.0092(K_{L,n} - 200)}]$$

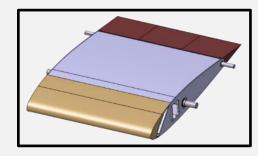




Experimental activities – Ice Roughness

- Objective: investigate influence of icing conditions on the characteristics of ice accretion roughness
- Experiments performed in TUBS Icing Wind Tunnel
 - HMDI airfoil: Span=0.5m, Chord=0.7m, non-symmetrical airfoil based on NASA CRM
 - Operating conditions encompassing both App.C and App. O

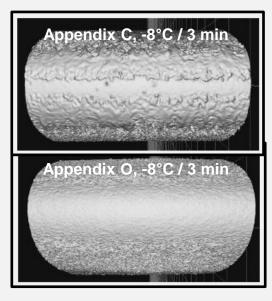
 $V_{air} = 40 \text{ m/s}$; $AoA = 0^{\circ}$; $T_{\infty} = -5 \text{ to } -16 \,^{\circ}C$; $Reynolds number Re \approx 2e6$ $MVD_{\infty} \approx 19 \mu m \text{ (App. C) } \& 70 \mu m \text{ (App. 0)}$; $LWC_{\infty} \approx 0.88 \text{g/m}^3 \text{ (App. C) } \& 0.56 \text{g/m}^3 \text{ (App. 0)}$ Accretion time t = 1.5, 3, 6, 9 min Experiments include several levels of $\eta_{f,0}$ Combinations of $\eta_{f,0}$ and A_c not investigated before Insight on effects of non-symmetric airfoils



Airfoil sketch



Iced airfoil in the test section

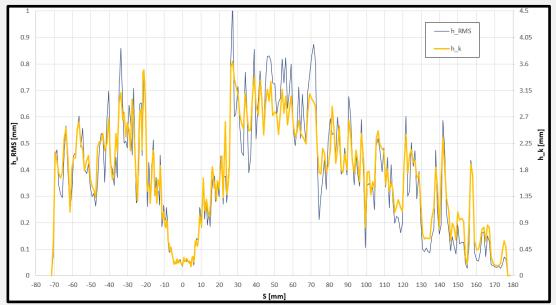


Accreted ice Front view

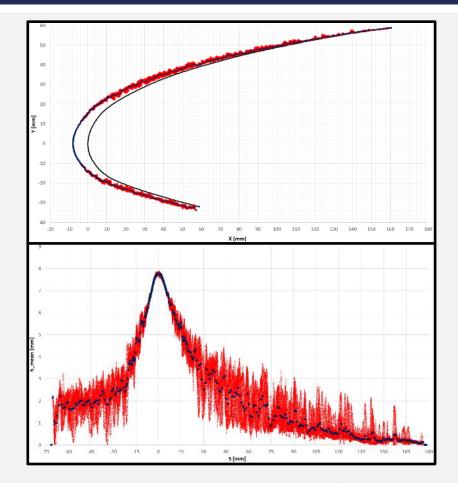


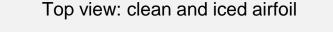
Ice Roughness – Experimental activities

- How to analyse the ice roughness?
 - Digitalization of the ice shape (photogrammetry method)
 - Development of tools for statistical analysis
 - Post-processing of the experimental data
 - Assessment of the tools



 h_{K} (orange) and h_{RMS} (blue) $\,$ vs. curvilinear abscissa



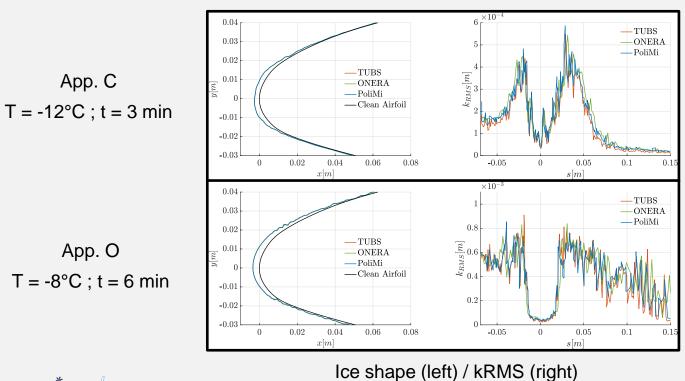


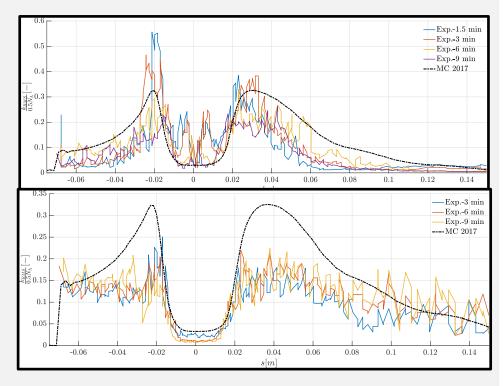
Bottom view: h_{mea}n (slices in red, mean in black dot)



Ice Roughness – Experimental activities

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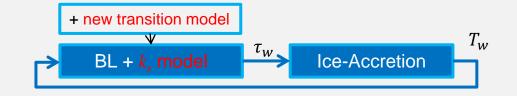


Rq comparison with McClain's model T= -8°C / Top view App.C / Bottom view App.O



Ice Roughness – Physical modelling

Model based on Fortin's bead model (roughness height) and Abu-Gahnnam & Shaw model (transition)



Roughness height model

Bead size: $e_b = \alpha_0 L R e_{\tau}^{\alpha_1} W e_{\tau}^{\alpha_2} H_1(\theta, \Delta \theta) H_2(R e_b)$

$$k_s = \alpha_3 e_b$$

Transition model

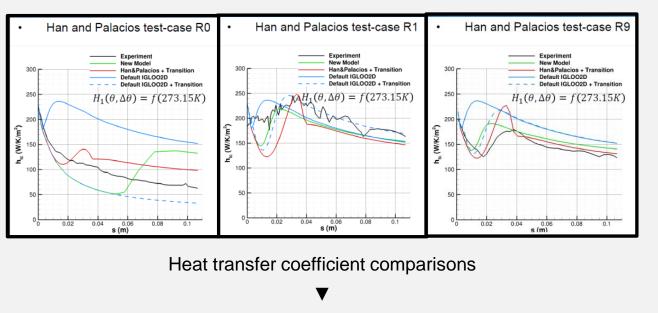
$$\gamma = 1 - e^{-\beta_1 \eta^{\beta_2}}$$

• Optimization process to determine the best values α_3 of and η using an IA approach Performed on 3 of the Han & Palacio's Heat Transfer Coefficient (htc) database

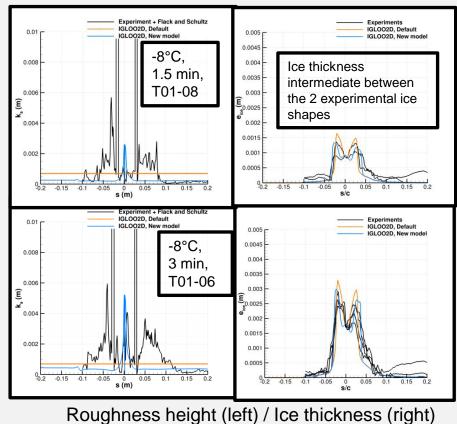


Physical modelling – Ice Roughness

- Assessment of the model based on
 - Roughness height characteristics (TUBS database)
 - Impact of the roughness height on measured data (htc Han & Palacio's database, ice shape TUBS database)



Good agreement for most of the cases



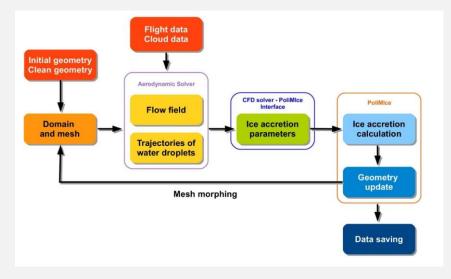
comparisons



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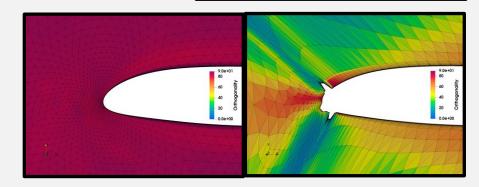
Numerical methods - Introduction

- Objective: develop efficient numerical methods to handle 3D simulations
 - Aiming at Predictor-Corrector or MultiStep approaches
 - Considering mesh adaptation to account for ice surface growth



A MANA

3D ice shape



But

Mesh displacement

- On physical grounds, 3D ice shape can be quite complex, 3D ice \neq 2D ice
- On numerical/modelling grounds, need to deal with mesh displacement, ice density modelling, mass conservation, ...



Numerical methods - Results

Several candidate methods have been developed by the partners:

- 3D Multi-step Immersed Boundary Method, Lagrangian displacement of surface mesh for ice accretion only (CIRA)
- 3D Predictor-Corrector plus remeshing, with Lagrangian displacement (ONERA)
- 3D Multi-step on conformal meshes with level-set and remeshing (POLIMI)
- 3D Multi-step on conformal meshes with Lagrangian (POLYMO)
- Very preliminary approaches for mass conservation are available
- P Definition of Numerical benchmark tests for T9.4
 - Baseline calculation: NACA23012 2D extruded cases (Ice Prediction Workshop database)

Run	V [m/s]	T [C]	P [Pa]	MVD [µm]	LWC [q/m³]	AoA [°]	Time [s]	Remarks
Case 241	103	-23°	92528	30	0.42	2°	300	Rime ice
Case 251	103	-12.6°	91700	21.5	1.64	2°	400	Monomodal SLD
Case 252	103	-12.6°	91700	21.5	1.64	2°	400	Bimodal SLD

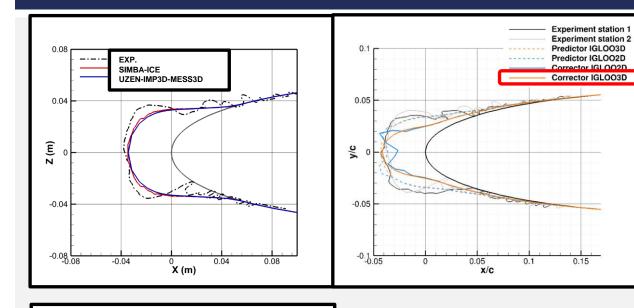
Benchmark tests: 30° swept NACA0012 (Ice Prediction Workshop database)

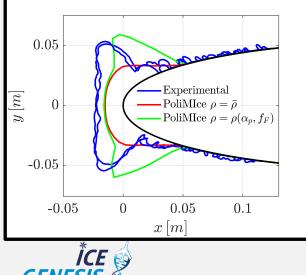
Case 361	103	-16°	92321	34.7	0.5	0°	1200	Rime ice	
Case 362	103	-7°	92321	34.7	0.5	0°	1200	Glaze ice	





Numerical methods – Results





Case 362 – Glaze ice (POLIMI)

0.15

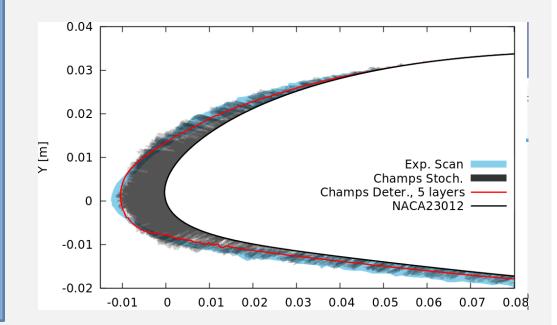
- Less correctly captured in 3D
- Ice density plays a role

Case 361 – Rime ice (CIRA, ONERA)

Correctly captured in 3D

Case 241 – Rime ice (POLYMO)

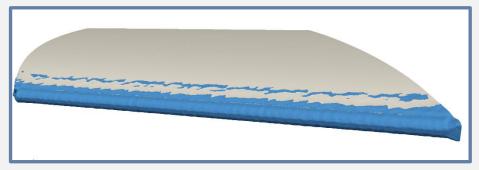
• Stochastic approach vs. Deterministic approach



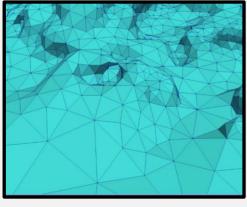
Numerical methods – Some more results

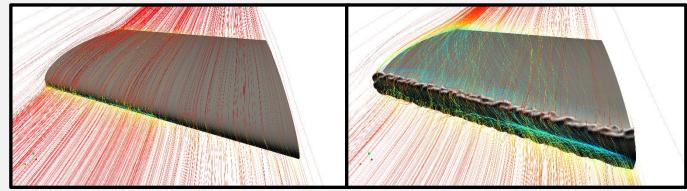
Observations

- Physics of 3D ice accretion results in non-negligible numerical difficulties
- Unsteady ice accretion is important to describe the whole process
- Models are still not satisfactory (e.g. ice density)



Multi-connected ice shapes (ice in blue)





Collection efficiency coefficient: single step (left), multistep (right)

Mesh issues with very refined grid leading very small ice structures



CO 16

Conclusions & Perspectives

- Main achievements on the experimental and modelling parts so far
 - Academic experiments performed on two important topics for SLD: drop impact and roughness
 - Roughness
 - Experimental methodology clearly defined
 - Ongoing activity to build a model to account for roughness & transition
 - Drop impact
 - o Insights gained thanks to some experiments or some complex/basic models but...
 - o Improvements of the existing models are not that conclusive
 - New phenomena to be possibly investigated (Dendritically Frozen Drop, erosion, drop deformation)
 - High-altitude effects to be investigated
- Main achievements on the numerical part so far
 - Extension of the capabilities of the 3D tools ongoing (Predictor-Corrector, Multistep)







THANK YOU



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